



# Collision Avoidance Operations at DLR/GSOC

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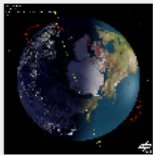
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# Introduction

- GSOC collision avoidance system (COLA) operational since 2009
- Monitoring satellites 5 in LEO, 2 in GEO:  
TerraSAR-X and TanDEM-X (514 km), GRACE-1&2 (460 km), PRISMA (750 km)
- Input: **TLE** and **Conjunction Summary Message** from JSpOC
- Daily proximity prediction using TLEs (for 7 days) running in automated process
- Automated warning and analysis process available

## COLA FD Web



### General

- [FD Ops Home](#)
- [Overview](#)

### Available Products

- [GRACE-1](#)

DLR

German Space Operations Center

Program COLA run on 2011/02/22 02:01

PREDICTION FROM: 2011/02/22 02:01:47.000

TO : 2011/03/01 02:01:47.000

REL.DIST < 10.0 [km]

RADIAL\_DIST < 3.0 [km]

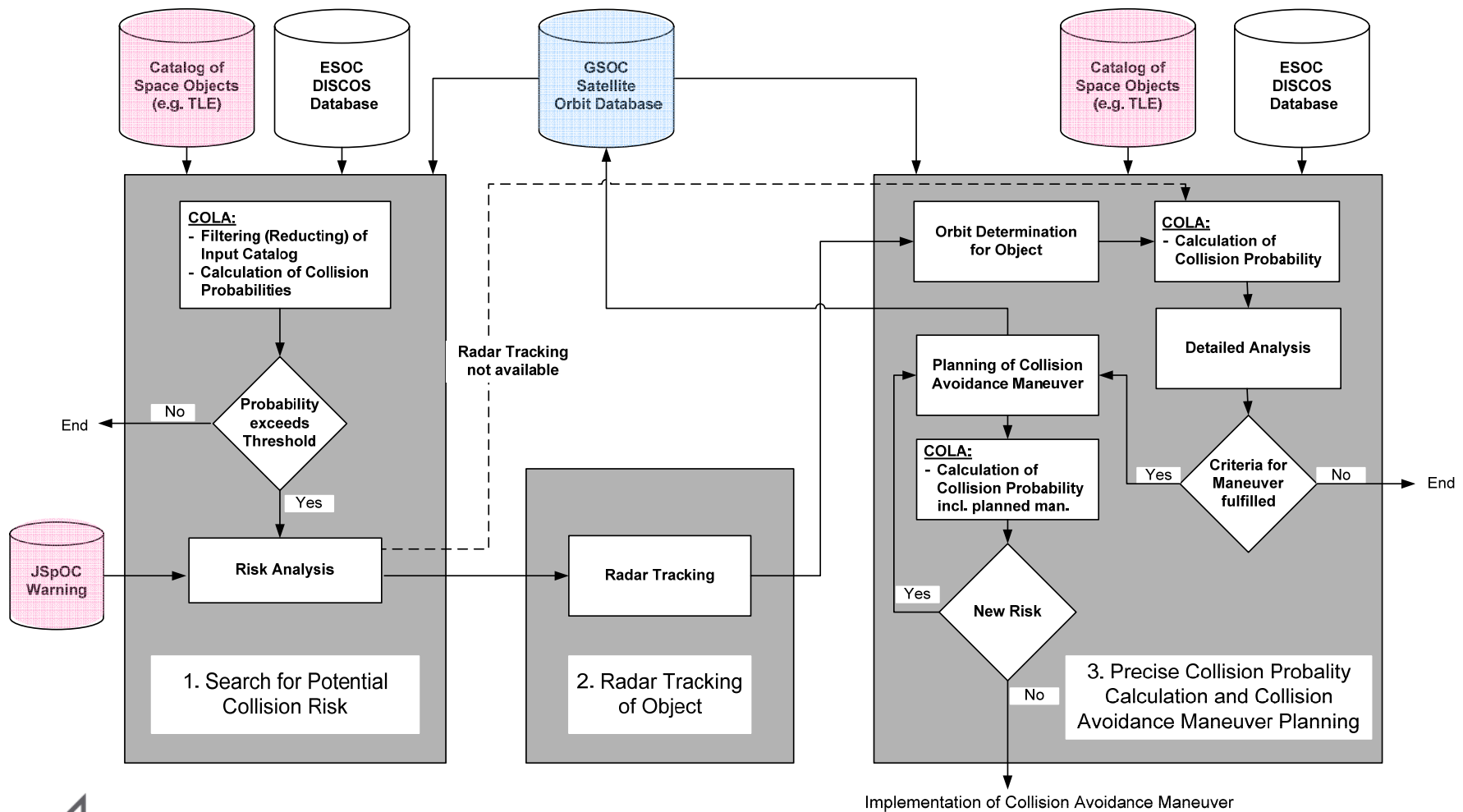
TARGET\_R : 2.60 [m]

OBJECT\_R : 2.00 [m] (Default)

SatID	Name	Days since	Time of approach	Max.Prob	Obj.R	1 Sigma in RTN			
				Min.Range	Rel.Vel				
				[km]	[km/s]	OrbPl.Angl			
						[deg]			
				R	T	N			
				[km]	[km]	[km]			
				OrbArcDist	TimFromNode	DstFromNode			
				[km]	[sec]	[km]			
31698	TerraSAR-X	0.528	S:2011/02/22 06:32:46.429	2.73E-07		DEFAULT	0.002	0.124	0.001
27134	PSLV DEB	1.537	2011/02/22 06:32:46.792	8.436	14.796	152.91	0.204	2.315	0.385
		0.188	E:2011/02/22 06:32:47.155	1.476	-1.945	-8.074			
				1.427	2.338	17.728			



# GSOC Collision Avoidance Process Overview





# Automated Warning

Automated warning based on

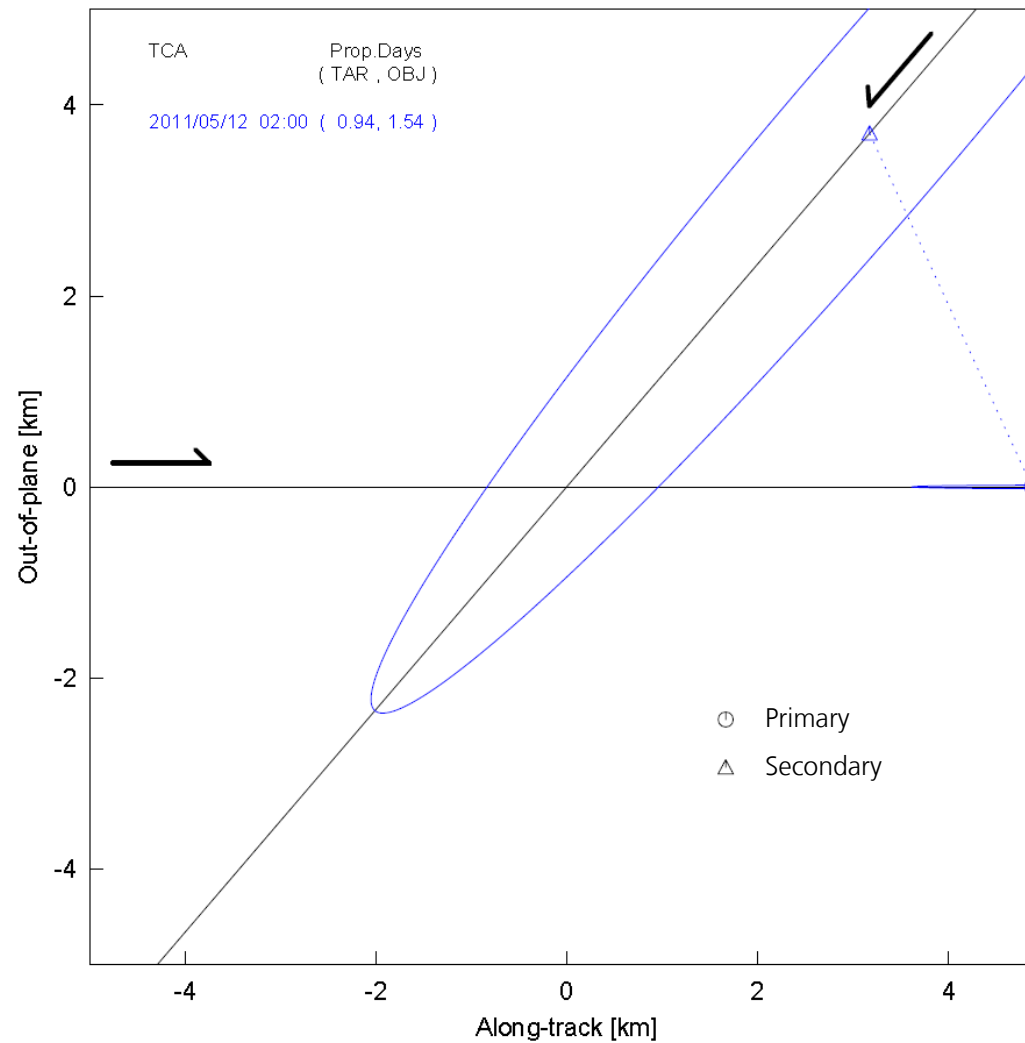
- Daily prediction using TLEs (maximum probability  $> 10E-04$  and radial distance  $< 300$  m)
- Notification from JSpOC

The latest COLA prediction results are sent to FD staff per E-mail

- Prediction summary (current and history)
- CSM summary (in case of JSpOC notification)
- First analysis results
  - Close approach geometry (example see next slide)
  - TLE statistics of the secondary object (subsequent 2 slides)



## Example of Close Approach Geometry



11-MAY-2011 07:01

# Example of TLE Statistics (1/2)

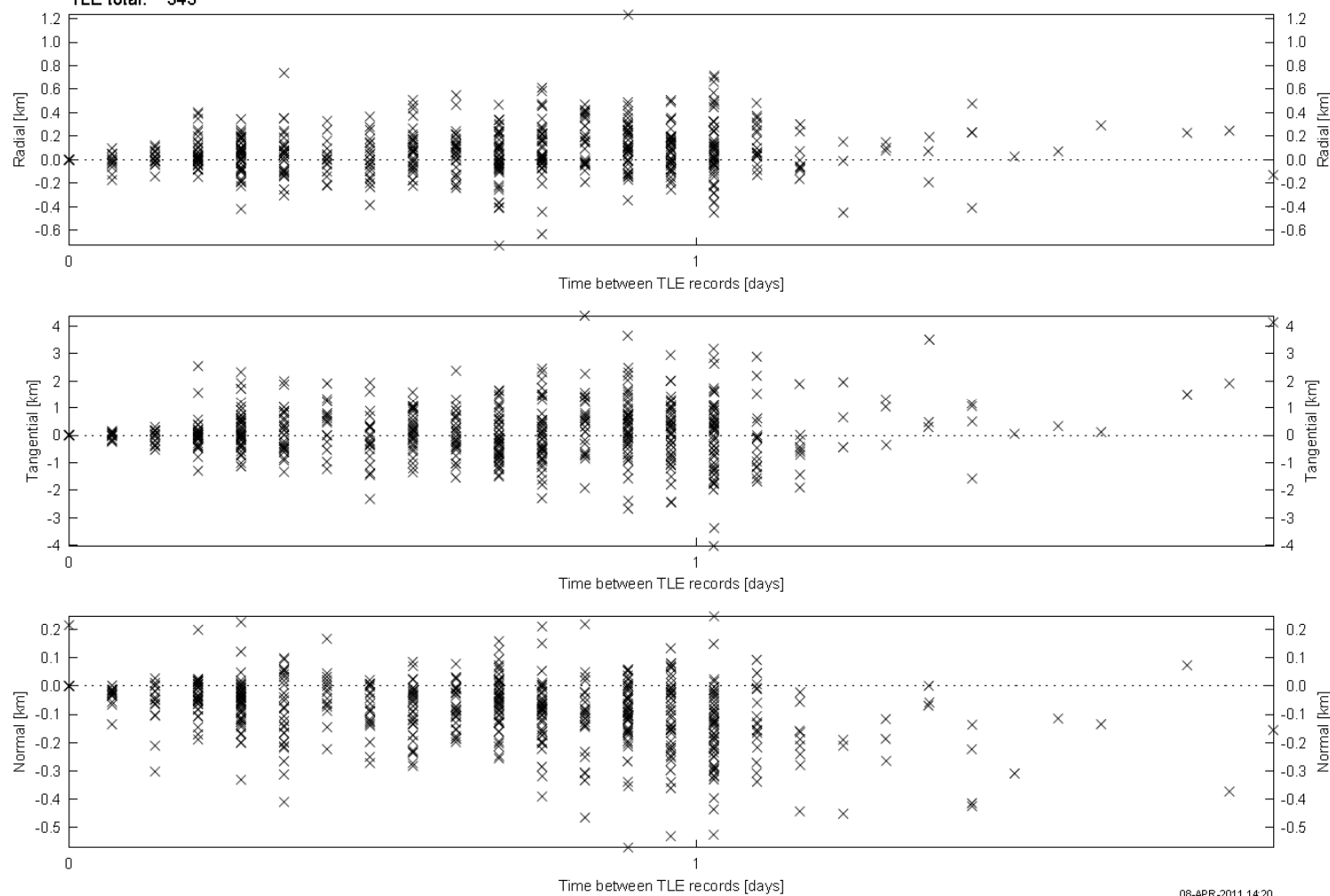
## TLE-RTN statistics for NORAD object #04941 - Prop. to the next TLE

Covered period: last 364 days

Mean, RTN : 0.071 0.121 -0.091

TLE total: 545

Sigma, RTN : 0.202 1.018 0.116





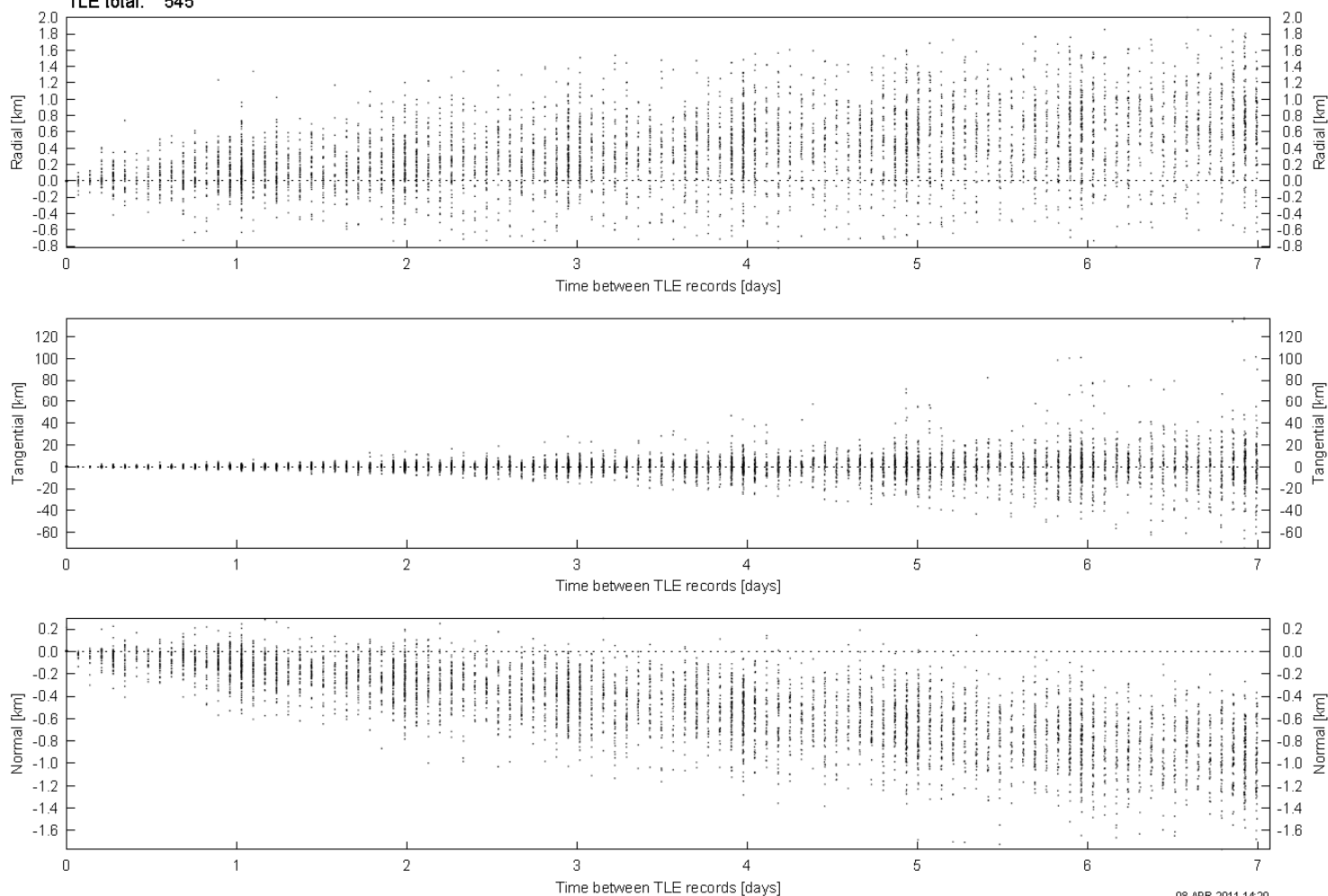


## Example of TLE Statistics (2/2)

TLE-RTN statistics for NORAD object #04941 - Prop. to TLEs up to next 7d

Covered period: last 364 days

TLE total: 545

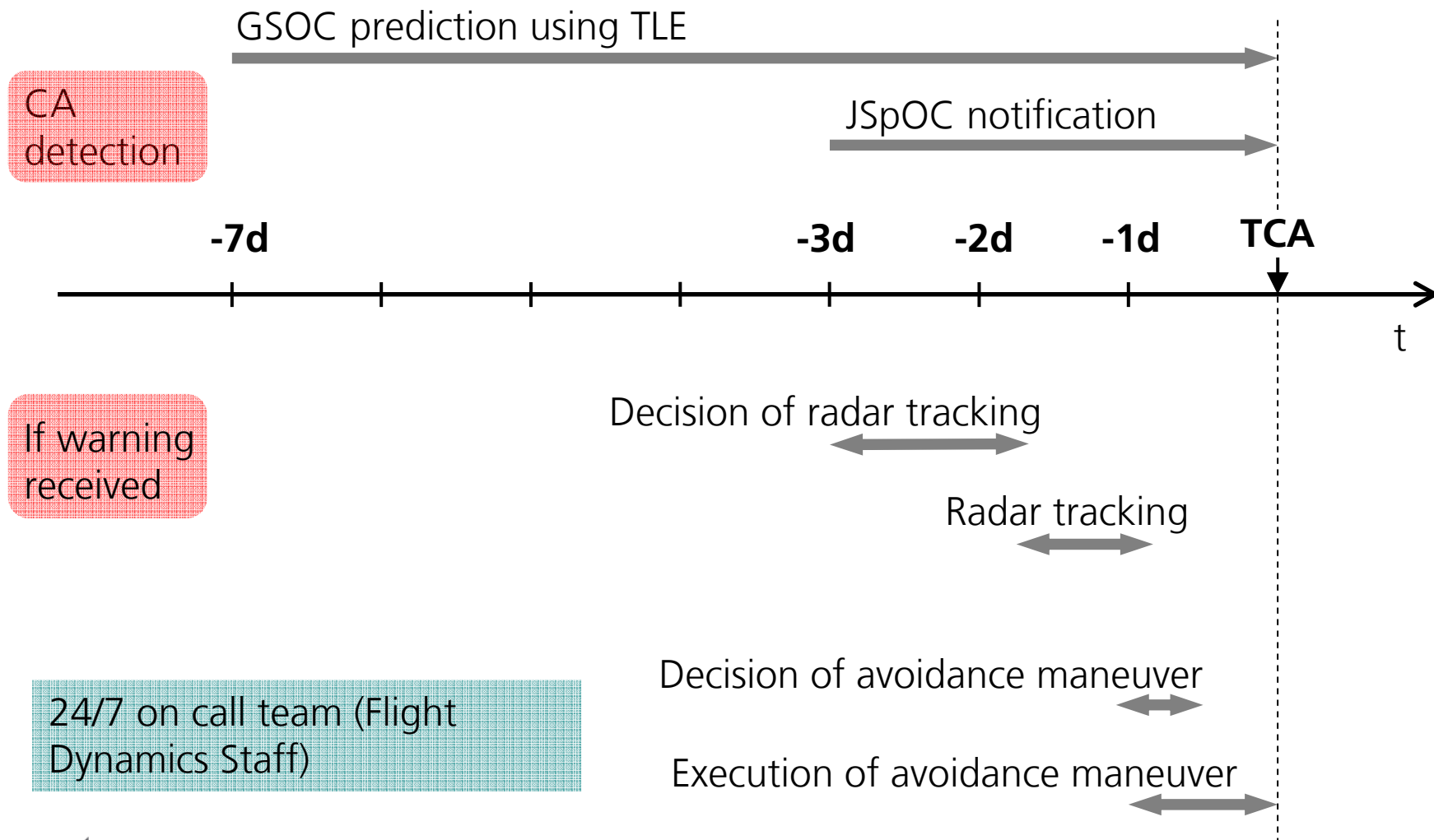


08-APR-2011 14:20





# Conjunction Avoidance Timeline





# Maneuver Strategy for TerraSAR-X and TanDEM-X (1/3)

## Mission Characteristics

- TerraSAR-X (TSX, launch 2007), TanDEM-X (TDX, launch 2010)
- Radar Satellites flying in a close formation (with TSX as Reference)
- TSX is controlled against a reference orbit inside a tube of 250 m radius
- TDX is flying in a close formation with TSX in a relative distance  $< 500$  m
- Tube control currently requires 1 orbit control maneuver ( $\sim 2$  cm/s) per day (planned for TSX and duplicated with TDX)
- Number of Tube control maneuvers is increasing with higher solar activity
- Formation control requires 2 maneuvers ( $\sim 4$  mm/s) per day executed by TDX only





## Maneuver Strategy for TerraSAR-X and TanDEM-X (2/3)

### TSX/TDX Collision Avoidance Process

- Tube and formation flight requirements shall be violated to a minimum (tight 3 years measurement schedule)
- Investigation of scenarios for shifting planned maneuvers to avoid additional maneuvers
- Analysis of these scenarios with respect to
  - Safe collision avoidance
  - Compliance of tube and formation keeping requirements
  - Compliance with observation plan (in strong coordination with mission planning and mission operations)
  - Compliance with ground station availability (for uploading of maneuver commands)
- Implementation of the selected scenario (including modified observation plan, if necessary)
- In case of violation of tube and formation requirements a recovery operations is to plan and implement after execution of collision avoidance maneuver



# Maneuver Strategy for TerraSAR-X and TanDEM-X (3/3)

## TSX/TDX Collision Avoidance Maneuver Strategy

- TSX collision avoidance
  - Change execution time/size of regular maneuver, TDX replicates the maneuver
  - TSX perform two maneuvers (for collision avoidance and for re-acquisition of reference orbit)
    - TDX replicates the maneuvers (fuel-expensive)
    - TDX remains passive → formation has to be re-acquired afterwards (time-consuming)
- TDX collision avoidance
  - TDX performs several maneuvers for collision avoidance and formation re-acquisition
  - TSX remains passive

# TerraSAR-X Close Approach to Pegasus Debris (1/3)

TCA	[UTC]	2010/08/07 13:19:35
Object		Pegasus debris (ID 24978)
Estimated size		~10 cm (RCS: 0.010 m <sup>2</sup> )
Perigee/apogee of object	[km]	464 / 1181 (e:0.044, i: 81.5°)
Relative velocity	[km/s]	15.1
Orbital plane angle	[deg]	160

- Warning from JSpOC, TCA-1.5 days
- Detected also by the daily (TLE) monitoring since TCA-7days (max Prob. >1.0E-04)

## Close Approach Prediction

	JSpOC warning (TCA-1.5d)	GSOC analysis (TCA-1.5d)	Daily prediction (TCA-1d)
TSX orbit	JSpOC	TSX precise	TSX precise
Debris orbit	JSpOC	JSpOC	TLE
Probability	N/A	3.4E-04	1.4E-05
Min.distance [m]	90	<b>81</b>	1064
Rel.distance (RTN) [m]	(69; -7; -58)	( <b>71</b> ; 5; 39)	(166; -184; -1035)



PEGASUS DEBRIS

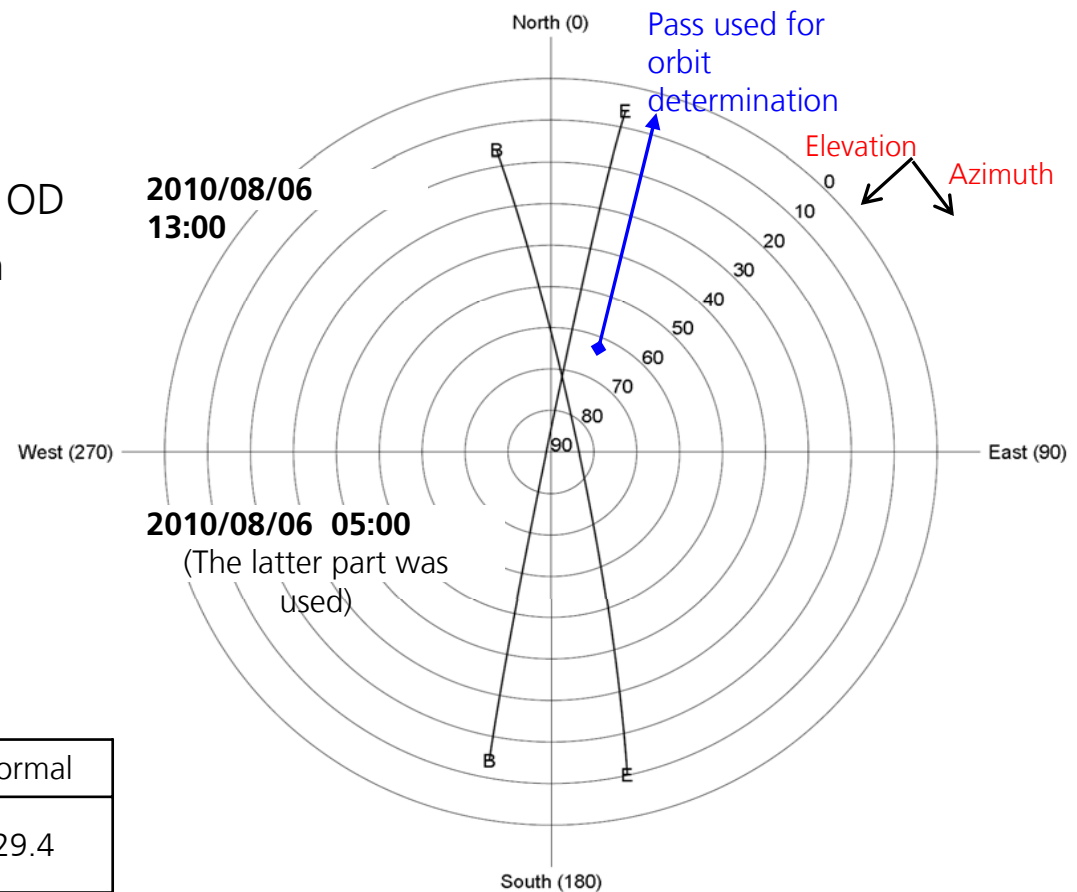
TerraSAR-X



## TerraSAR-X Close Approach to Pegasus Debris (2/3)

Orbit refinement using TIRA radar

- 1.5-1.0 day before TCA
- Data arc of ~10 hours was used for OD
- Accuracy improvement, especially in radial direction of 2.5 m



Orbit Error at TCA (1-sigma, in [m])

	Radial	Along-track	Normal
Radar tracking (TCA - 1.0 d)	2.5	16.3	29.4
JSpOC (TCA - 1.5 d)	12	135	18





## TerraSAR-X Close Approach to Pegasus Debris (3/3)

After the radar tracking, a collision avoidance maneuver was decided

- Increase of radial separation ~150 m
- 2 maneuvers, for collision avoidance and coming back a half orbit before/after TCA
- Each maneuver was ~5 cm/s and ~64 g hydrazine was used in total

Latest Prediction with / without Maneuver

	FHR tracking (TCA-1d)	Incl. maneuver (TCA-0.8d)
TSX orbit	TSX precise	TSX precise
Debris orbit	Radar tracking	Radar tracking
Probability	1.1E-03	1.7E-05
Min.distance [m]	216	337
Rel.distance (RTN) [m]	(-19; -31; -213)	(-165; -46; -291)
Orbital arc distance [m]	13	174



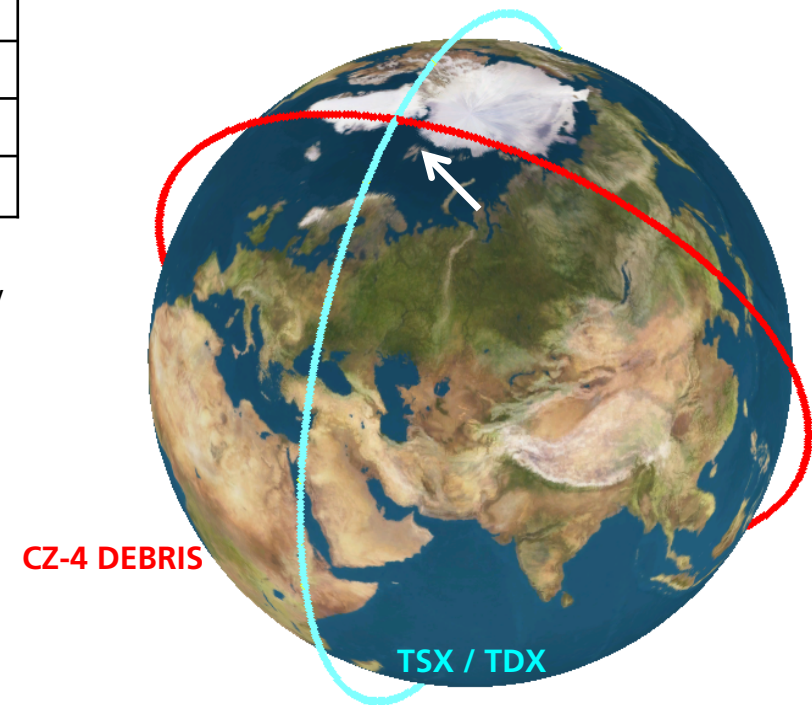
# TSX/TDX Close Approach to CZ-4 Debris (1/4)

TCA	[UTC]	2011/03/25 15:08:11
Object		CZ-4 debris (ID 26278)
Estimated size		~15 cm from RCS
Perigee/apogee of object	[km]	490 / 506 (e:0.001, i: 98.5°)
Relative velocity	[km/s]	10.8
Orbital plane angle	[deg]	90

- Warnings from JSpOC, TCA-3 days and -0.5 day
- No Pre-warning from TLEs
- Radar tracking was not available

	<b>TDX</b>	<b>TSX</b>
Primary orbit	TDX precise	TSX precise
Secondary orbit	JSpOC	JSpOC
Probability	2.8E-05	6.8E-06
Min.distance [m]	<b>88</b>	186
Rel.distance (RTN) [m]	(-87; -9; -9)	(173; -50; -50)
1-sigma primary [m]	(6; 191; 5)	(2; 69; 1)
1-sigma secondary [m]	(20; 876; 11)	(20; 876; 11)

First critical CA,  
since TSX-TDX formation was achieved



Prediction results by GSOC



## TSX/TDX Close Approach to CZ-4 Debris (2/4)

A collision avoidance maneuver was decided for TanDEM-X

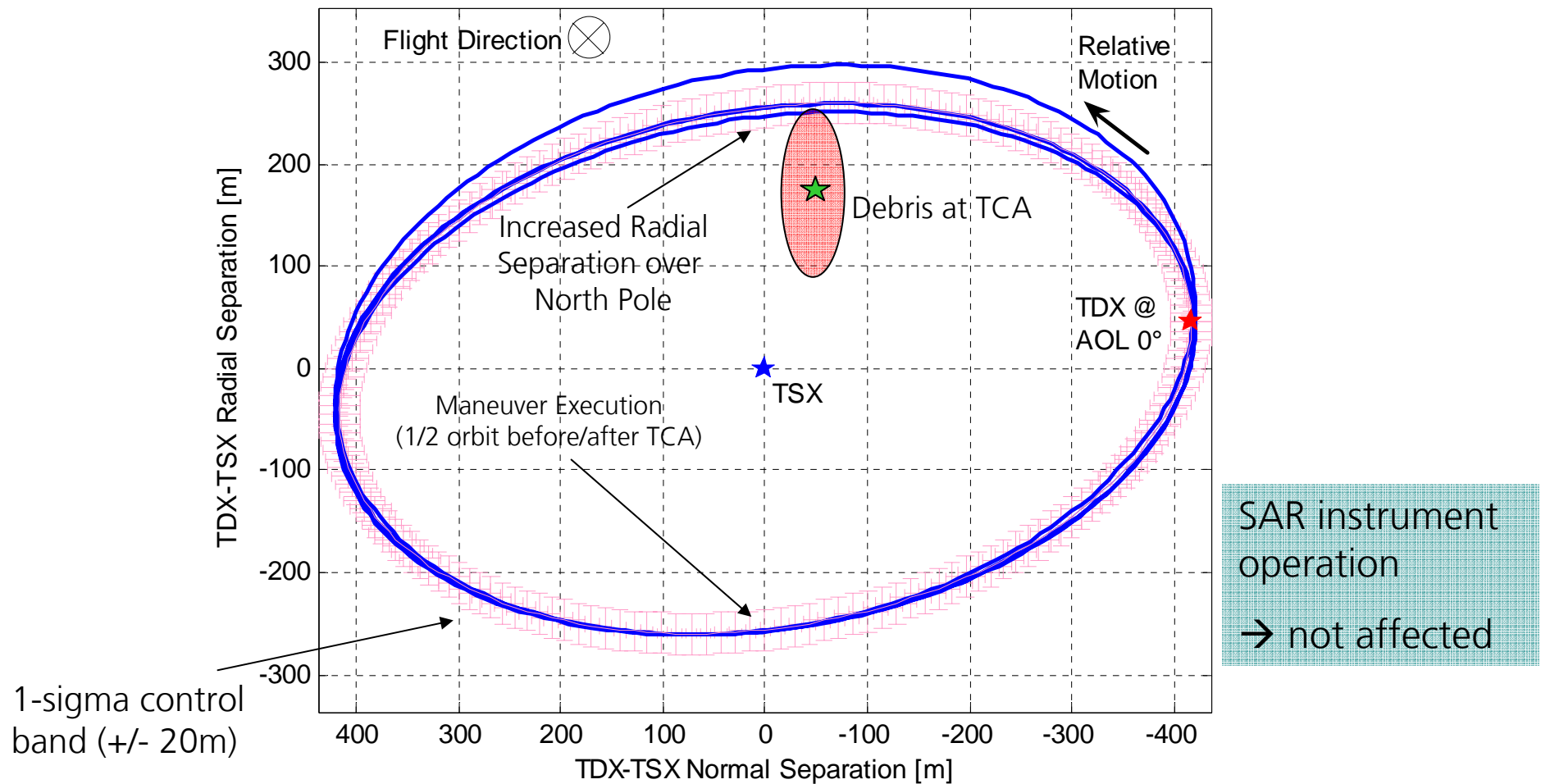
- Increase of radial separation ~40 m
- 2 maneuvers, for collision avoidance and coming back a half orbit before/after TCA
- Each maneuver was ~1 cm/s and ~14 g hydrazine was used in total

Latest Prediction with / without maneuver

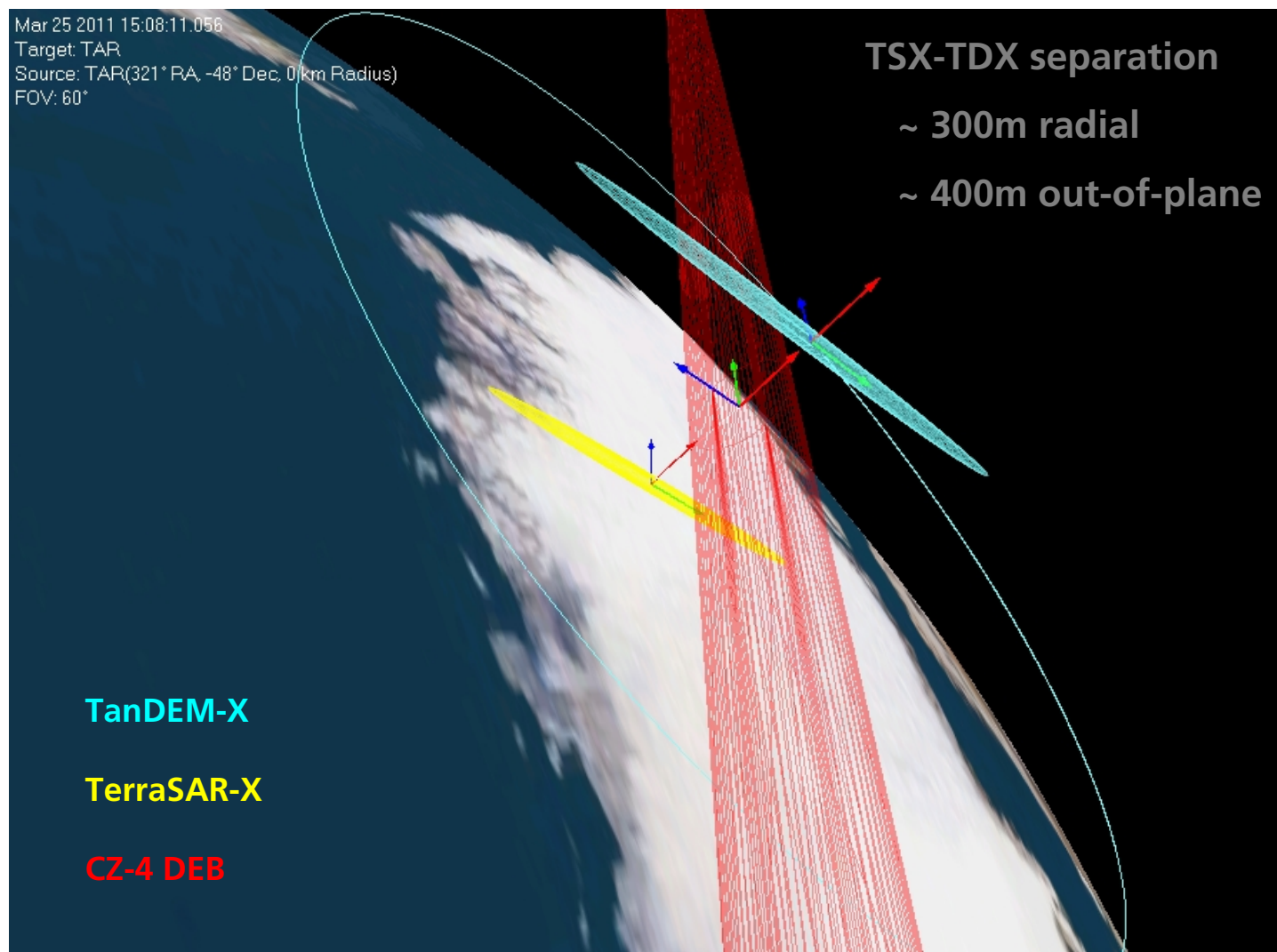
	without maneuver (TCA-0.4d)	Incl. maneuver (TCA-0.4d)
TSX orbit	TDX precise	TDX precise
Debris orbit	JSpOC	JSpOC
Probability	2.8E-05	1.3E-05
Min.distance [m]	88	136
Rel.distance (RTN) [m]	(-87; -9; -9)	(-125; 38; 38)
Orbital arc distance [m]	87	125

# TSX/TDX Close Approach to CZ-4 Debris (3/4)

TDX-TSX Relative Motion



## TSX/TDX Close Approach to CZ-4 Debris (4/4)





# Statistics of Events

Number of Handled Close Approaches (as of January 2011)

	Altitude	Record since	Analyzed	JSpOC warning	Maneuver
TerraSAR-X	514 km	Aug. 2009-	12	5	3
TanDEM-X	514 km	Jun. 2010-	7	3	2
GRACE-1	460 km	Nov. 2009-	3	3	0
GRACE-2	460 km	Nov. 2009-	0	0	0
PRISMA	750 km	Mar. 2011-	2	2	0



## Future Work

Enhancement of the System

- Further Analysis Tools
- Further Automation of Process
- Geostationary Missions (Orbit Determination based on Optical Measurements)